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# High-accuracy optical measurement of flatness for large objects

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## Abstract

A high-accuracy non-contact optomechanical system has been designed for measuring the surface profile of relatively flat and large objects. The experimental set-up consists of a motorized gantry, a rangefinder, a CCD chip and a laser diode. This set-up permits discrete measurements to be performed on objects with a maximum plane surface area of  $2.6 \times 0.5 \text{ m}^2$  along both the  $X$  and  $Y$  axes. Experiments were carried out on carbon sandwich panels. An uncertainty of  $\pm 8 \text{ }\mu\text{m}$  has been obtained on flat and smooth surfaces; a  $\pm 30 \text{ }\mu\text{m}$  uncertainty has been determined for a rough carbon sandwich panel.

**Keywords:** shape measurement, laser sensor, non-destructive testing

## 1. Introduction

This paper describes an alternative method of shape measurement for large and relatively flat objects. This application forms part of an important stage of a wider project (ALICE), dedicated to fundamental research in particle physics [1]. It requires the building of a large detector system, the structural elements of which are carbon sandwich panels. Controlling the flatness of these large panels ( $2450 \text{ mm} \times 450 \text{ mm}$ ) is thus essential.

Several methods exist for object profile measurements. For instance optical techniques [2] such as holographic interferometry [3], shearography [4], moiré topography [5, 6] or the fringe projection method [7] are widely used. These techniques permit non-contact, full-field measurements and fast data acquisition but provide high accuracy only for small objects.

In parallel, numerous papers have also been published on high-accuracy measurement of large optical surfaces ( $>1 \text{ m}$  in diameter). In most cases, a contact probe [8, 9] or optical probe [10, 11] is used. As an example, systems have been developed to measure the shape of x-ray telescope mirrors [10] or of the flat table of a machine tool [12].

However, the cost of current surface profiling sensors, whether optically or non-optically based, is prohibitive. To overcome this constraint we have therefore decided to develop

our own profilometer to meet our measurement requirements with various user-defined parameters. In this paper we present the description of a 3D measuring system of relatively low cost with a high resolution over the entire measured surface. The application of our set-up has been demonstrated on a  $2450 \text{ mm} \times 450 \text{ mm}$  carbon panel.

## 2. Experimental set-up

### 2.1. System description

The experimental set-up consists of a motorized gantry, built to our specifications in our laboratory and customized for profiling relatively flat surfaces. Each axis of the gantry is controlled by a step motor via an individual cable. This set-up permits measurements to be performed on objects with a maximum plane surface area of  $2.6 \times 0.5 \text{ m}^2$  along both the  $X$  and  $Y$  axes. On the vertical rail, a mechanical fixture has been added to allow both a laser triangulation rangefinder and a CCD chip to be held rigidly. The schematic diagram of the system is given in figure 1. The carbon panels have to be held vertically in order to overcome the problem of deformation under their own weight. Each panel is thus placed vertically, parallel to the  $X$ -axis, and is held in place by a simple suction system.

At each  $(X, Y)$  displacement along the carbon panel, the absolute distance between the plate and the rangefinder is first

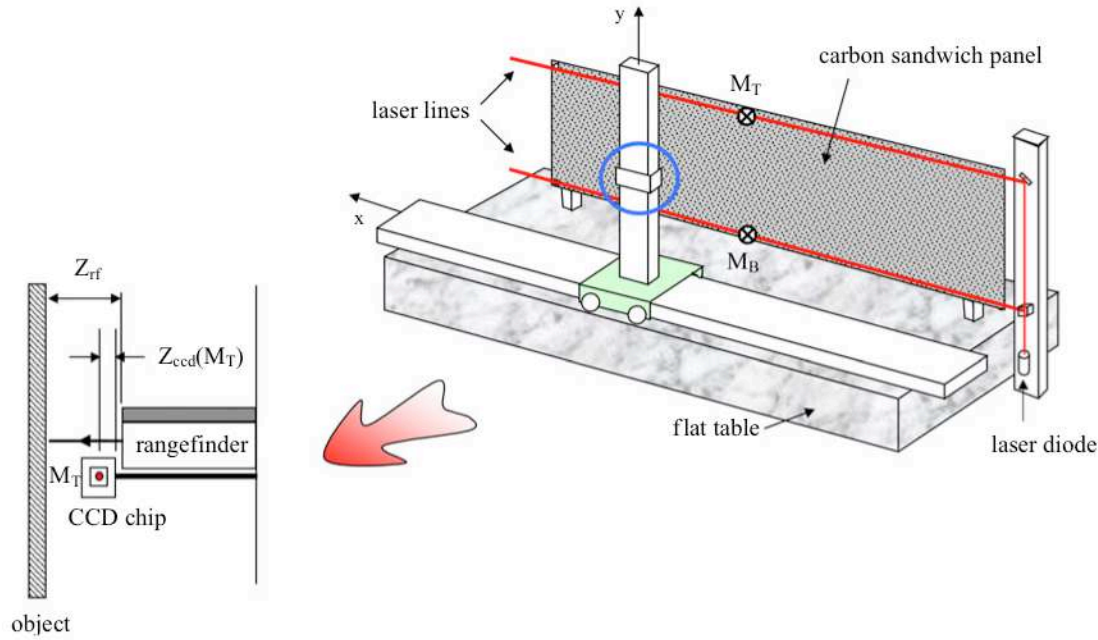


Figure 1. Schematic of the experimental set-up.

measured by the rangefinder. This direct measurement is, however, not sufficient to determine the object profile since it uses the displacement along the  $X$  rail as a mechanical reference. Although the maximum curvature of the  $X$  rail has been measured to be within  $200 \mu\text{m}$ , an uncertainty of  $\pm 30 \mu\text{m}$  is required on such large surfaces. The use of this mechanical reference ( $X$ -rail displacement) as flatness data is, thus, not appropriate as it is impossible to know the exact position of this fixture during measurements. The measurement must be calibrated from an external reference plane. We thus designed a calibration system to correct these errors.

## 2.2. Calibration

Initially, a reference plane, exploiting a laser plane generator, was introduced vertically between the gantry and the panel. The rangefinder is displaced in front of the surface to be measured. Its position is monitored relative to the reference laser plane by the use of a CCD chip, which intercepts this plane. The height of each point is then calculated as

$$Z = Z_{\text{rf}} - Z_{\text{CCD}} \quad (1)$$

where  $Z_{\text{rf}}$  and  $Z_{\text{CCD}}$  are the rangefinder measurement and the laser plane position on the CCD chip, respectively. A surface profile can thus be obtained in this way.

Although the use of an internal laser reference plane seemed to be a very simple solution to the reference problem, it was later found that the laser plane was finally not so flat. The laser line generator employed was a commercial, high-precision model, consisting of a prismatic optical head illuminated by a laser diode spot. Measurements with this laser line led to erroneous results. A thorough investigation revealed two phenomena: optical distortion due to the prismatic head, which bends the beam, and, moreover, some interference fringes which modify the transverse intensity profile of the beam at several locations.

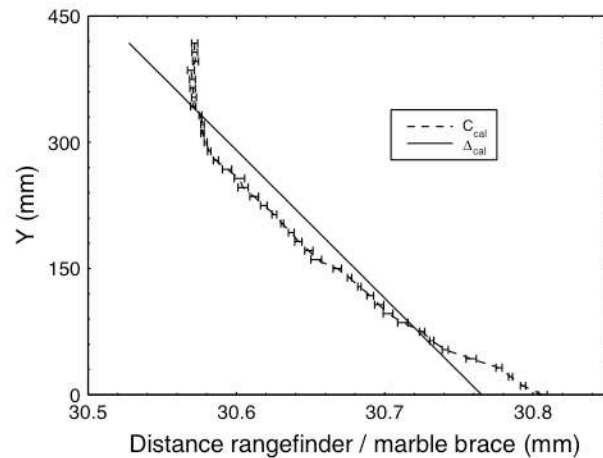


Figure 2.  $Y$ -axis calibration curve.

We therefore designed another reference plane, based on the use of a high-precision vertical mechanical rail. A calibration of the fixture displacement along this vertical rail enables us to obtain a highly reproducible displacement. In order to obtain this slightly curved movement (which is defined as  $C_{\text{cal}}$ ) along the  $Y$ -axis, approximately 50 measurements have been realized on the same reference surface (high-precision marble brace, 50 cm high, quoted flatness  $\pm 2 \mu\text{m}$ ). Figure 2 represents the mean experimental curve and the standard deviation (of  $\sim 4 \mu\text{m}$ ) associated with these measurements and the best-fit linear regression over measurements  $\Delta_{\text{cal}}$ . The maximum bending has been found to be  $60 \mu\text{m}$ . Moreover, tests have been realized (over 6 months) to check the validity of the calibration curve and the stability of our set-up. Taking into account the displacement consideration, we then conceived a reference plane based on the use of a twin laser line.

This reference plane consists of a laser diode, with anamorphic beam-conditioning optics, emitting in the visible







displacement along the  $X$  rail, associated with the two laser beams (top and bottom position). Then, we subtract from the bottom curve its best-fit linear regression and add to the resulting curve the best-fit linear regression of the top curve. With this correction, the point  $M_B$  becomes  $B$ . The two resulting curves, passing through  $M_T$  and  $B$ , create a perfect virtual reference plane  $P_{ref}$ .

It should be noted that it does not matter if this virtual plane is difficult to locate spatially as the flatness measurement is only a relative measurement.

The positioning of the beams and their adjustment for parallelism have been checked mechanically, with a maximal error of  $\pm 0.5$  mm along 2400 mm, i.e. an angular deviation of about  $10^{-2}$  deg. This influence of such an error on the surface profile measurement is of secondary importance compared with the accuracy of the rangefinder ( $20 \mu\text{m}$ ). As a matter of fact, the angular deviation between the two beams leads to a projection error (during the creation of the reference plane) which can be estimated to only one-hundredth of a micrometre.

The surface profile of the object is then deduced from the subtraction between the rangefinder data and the reference plane for each  $(X, Y)$  recorded point:

$$Z = Z_{rf} - Z_{ref} \quad (2)$$

where  $Z_{rf}$  is the distance between the rangefinder and the object, given directly by the rangefinder,  $Z_{ref}$  is the distance between the rangefinder and the reference plane data, to be determined. In the experimental configuration,  $Z_{ref} < 0$ . For any point  $A$ ,  $Z_{ref}$  is given by:

$$Z_{ref} = D_0 - Z_{cor} - Z_{cal} \quad (3)$$

where  $D_0$  is a constant mechanical distance (distance between CCD edge and the rangefinder). Exact knowledge of  $D_0$  is not necessary since we realize relative measurements.  $Z_{cor}$  is the distance between the reference plane (passing through  $M_T$  and  $B$ ) and the straight line passing through the origin of the CCD reference in the top and bottom position. The straight line represents the global slope of the vertical rail compared with the reference plane. For each  $X$  coordinate, the straight line is calculated from the pair of points  $(M_T, M_B)$ .

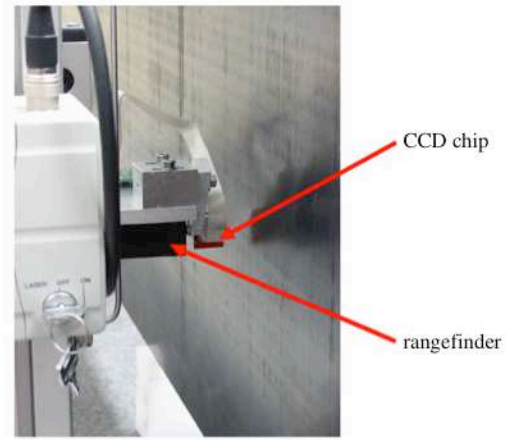
$Z_{cal}$  is the curve resulting from the subtraction between the rangefinder calibration curve along the  $Y$ -axis and its best-fit linear regression (figure 2):

$$Z_{cal} = C_{cal} - \Delta_{cal}. \quad (4)$$

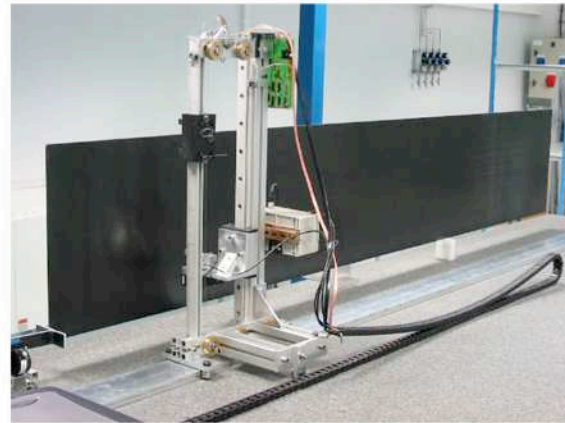
### 3. Results and discussion

Photographs of the operational system, built in our laboratory, are shown in figure 4. The vertical rail and fixture displacements, as well as the acquisition of rangefinder and CCD chip data, are entirely automated and controlled by a program written in Labview<sup>®</sup>. Processing of the recorded data is carried out via a Matlab<sup>®</sup> program and leads to the following important results: surface profile and flatness criterion.

The flatness criterion is directly derived from the profile map. Flatness is defined as the maximum distance between two perfect, parallel reference planes between which is placed the surface of interest. In order to guarantee relatively acceptable



(a)

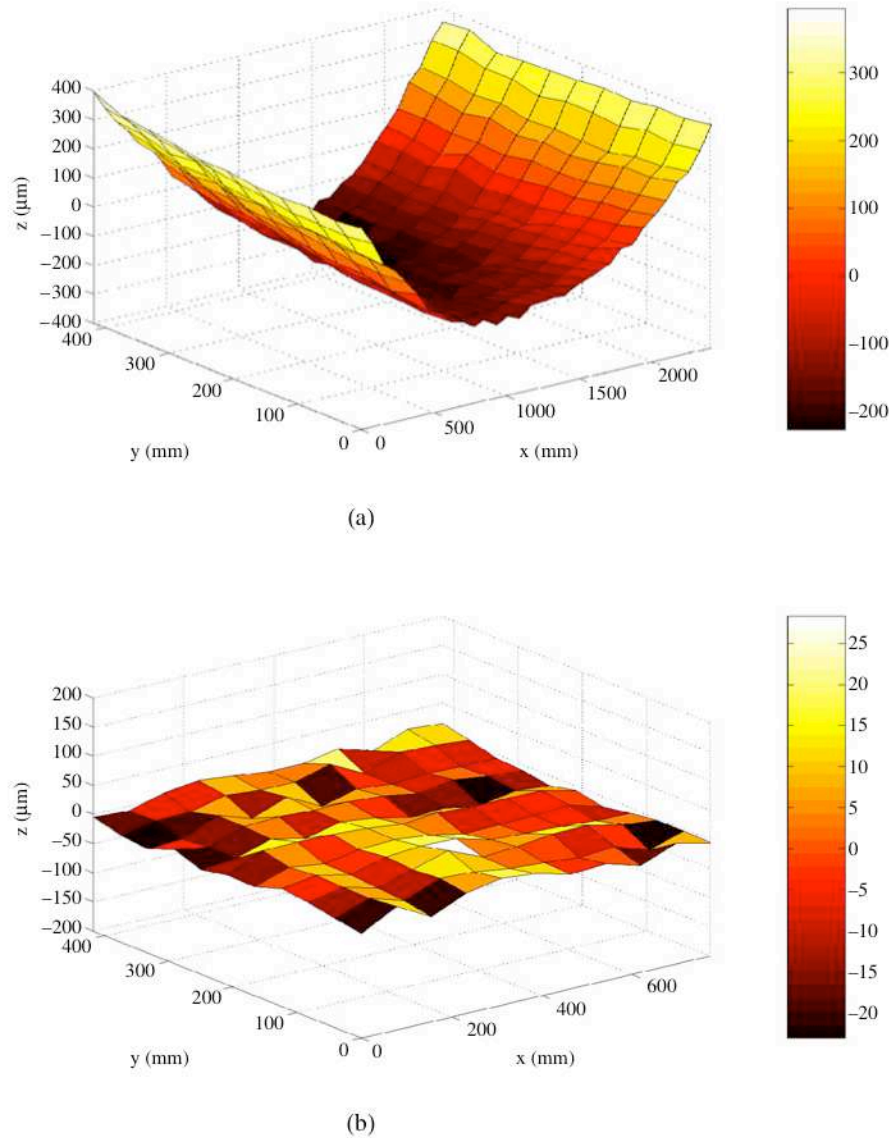


(b)

**Figure 4.** Photographs of the set-up: (a) lateral view, (b) global view.

results from the whole fundamental experiment of the ALICE project, and according to a conventional mechanical standard abacus, the flatness of these panels must be better than  $150 \mu\text{m m}^{-1}$ . The uncertainty of our profile mapping system has therefore been experimentally obtained at  $\pm 30 \mu\text{m}$  on the measured height of each point of the surface in order to achieve the overall flatness measurement uncertainty.

Preliminary experiments were carried out on carbon sandwich panels of dimension  $2450 \text{ mm} \times 450 \text{ mm}$ . Each panel is black and has a diffusive surface. As the laser rangefinder spot is  $50 \mu\text{m}$  in diameter, the state of roughness of the object surface is not important if the surface granularity is lower than the spot diameter—which is the case for carbon panels. Figure 5(a) shows the 3D contour map of a  $2450 \text{ mm} \times 450 \text{ mm}$  carbon panel. A relatively significant curvature along the  $X$ -axis can be observed. The calculated flatness is approximately  $400 \mu\text{m m}^{-1}$ , which is higher than the required tolerance. In fact, all the tests realized on  $2450 \text{ mm} \times 450 \text{ mm}$  panels showed a flatness greater than  $150 \mu\text{m m}^{-1}$ . A smaller and flatter carbon panel measuring  $850 \text{ mm} \times 450 \text{ mm}$  has also been studied and its 3D contour plotted in figure 5(b) for comparison, with a calculated flatness of  $150 \mu\text{m m}^{-1}$ . This is the required tolerance limit. The longitudinal curvature observed on the  $2450 \text{ mm} \times 450 \text{ mm}$  panels can thus be attributed to manufacturing faults.



**Figure 5.** (a) 3D map of a 2450 mm  $\times$  450 mm panel, (b) 3D map of an 850 mm  $\times$  450 mm panel.

In order to determine the uncertainty of the set-up, about 10 surface profile measurements have been realized on the same 2450 mm  $\times$  450 mm panel. On this panel, 11  $\times$  16 points were measured. The histogram shown in figure 6 represents the distribution of the measurement around the mean measured value. The experimental results present an uncertainty of  $\pm 30 \mu\text{m}$ , in an interval corresponding to 95% of the measurements, which is in good accordance with the previous specifications.

A similar experiment has been carried out on a smooth surface of smaller dimensions. The distribution histogram of the measurement around the mean value is represented in figure 7. The measurement uncertainty is  $\pm 8 \mu\text{m}$ , corresponding to 95% of the measurements.

Comparing these two results, the uncertainty of the set-up is mainly deteriorated by the slight non-reproducibility of the repositioning of  $X$  and  $Y$  ('re-initialization' of the mechanical fixture).

The current set-up, however, presents two other main drawbacks. Firstly, it is a time-consuming system ( $\sim 8$  min for

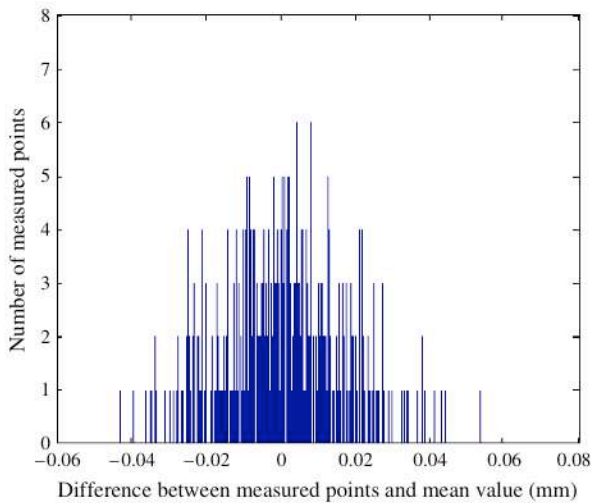
100 points). Secondly, only discrete measurements are allowed at present—the finest spatial resolution, in  $X$  and  $Y$ , which can be obtained being 0.8 mm. It should, however, be noted that higher spatial resolution will induce longer data acquisition times. A compromise must therefore be determined taking into account these two parameters.

To avoid these drawbacks, we are currently working on a new all-optical experimental configuration using the principle of the fringe projection method. A new algorithm is being developed, enabling us to reconstruct the profile of objects. Theoretically, the algorithm may be applied for profiling objects of any arbitrary size [13].

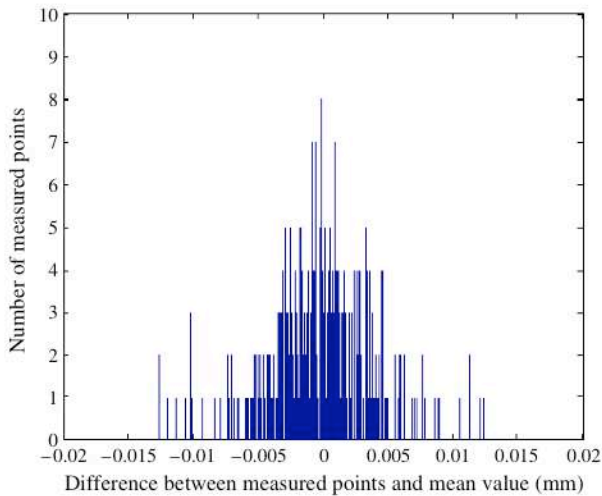
#### 4. Conclusions

We have designed a system for the measurement of any relatively flat surface profile. It allows non-contact, punctual, non-destructive measurements and can be employed on non-cooperative surfaces (diffusive and black surface of the carbon





**Figure 6.** Distribution histogram of distance measurement on 2450 mm  $\times$  450 mm carbon panel.



**Figure 7.** Distribution histogram of distance measurement on a flat surface.

panel for our application). Objects of surface areas up to  $2.6 \times 0.5 \text{ m}^2$  can be measured. The uncertainty of the set-up is  $\pm 30 \text{ }\mu\text{m}$  and the maximum deformation allowed for the

panel, due to the limitation of the rangefinder, is  $\pm 5 \text{ mm}$ . Furthermore, because of the disadvantage of the long time required, the spatial resolution must be chosen with care. The system is self-calibrated, fully automated and has a relatively low cost. This prototype can, moreover, be easily adapted to other specific needs.

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